# ministry of education and science of ukraine <br> National aerospace university "Kharkiv Aviation Institute" <br> Department of aircraft strength 

Course
Mechanics of materials and structures
HOME PROBLEM 12
Stress Analysis in Eccentric Tension - Compression

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National aerospace university
"Kharkiv Aviation Institute"
Department of aircraft strength
Subject: mechanics of materials
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Topic: stress analysis in eccentric tension - compression
Full name of the student, group

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Complexity: 1


Given: $I$-beam №(18), $\mathrm{F}=10 \mathrm{kN},[\sigma]=160 \mathrm{MPa}$.

Goal: 1) calculate stresses in an arbitrary cross - section $A B C D$ and check the strength; 2) Draw the graph of stress distribution in cross - section $A B C D$; 3 ) determine analytically position of neutral axis.

Full name of the lecturer
signature

Mark: $\square$

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Data:

1) Column is I-beam N18 with the following geometrical properties: $\quad h=18 \times 10^{-2} \mathrm{~m}, \quad b=9 \times 10^{-2} \mathrm{~m}$, $t=8.1 \times 10^{-3} \mathrm{~m} \quad$ (web), $\quad d=10^{-3} \mathrm{~m} \quad$ (flange), $A=23.4 \times 10^{-4} \mathrm{~m}^{2}, \quad I_{y}=82.6 \times 10^{-8} \mathrm{~m}^{4}$, $I_{z}=1290 \times 10^{-8} \mathrm{~m}^{4}, \quad W_{y}=18.4 \times 10^{-6} \mathrm{~m}^{3}$, $W_{z}=143 \times 10^{-6} \mathrm{~m}^{3}$
2) External load $F=10 \mathrm{kN}$.
3) Allowable stresses $[\sigma]_{t}=20 \mathrm{MPa},[\sigma]_{c}=80 \mathrm{MPa}$.

## Goal:

1) Determine the internal forces in an arbitrary cross section $A B C D$ far from the Saint-Venant zone.
2) Calculate acting stresses in potentially critical points of an arbitrary $A B C D$ cross section.
3) Design the graph of stress distribution over the contour of $A B C D$ section and determine the neutral axis position.
4) Determine the neutral axis position analytically.

Fig. 1

## Solution

1) Let us cut the column mentally in an arbitrary $A B C D$ section according to the method of sections and determine internal forces in top and bottom cross sections of the column (see. Fig. 2):

$$
\left|N_{x}\right|=F=10 \mathrm{kN}, \quad\left|M_{y}\right|=F \frac{b}{2}=450 \mathrm{Nm}, \quad\left|M_{z}\right|=F\left(\frac{h}{2}-d\right)=819 \mathrm{Nm} .
$$

2) Calculate the stresses in potentially critical angular points of $A B C D$ section. Calculate preliminary the stress moduli generated by three internal forces:
$\left|\sigma\left(N_{x}\right)\right|=\frac{10 \times 10^{3}}{23.4 \times 10^{-4}}=4.27 \mathrm{MPa}$,
$\left|\sigma_{\max }\left(M_{y}\right)\right|=\frac{M_{y}}{W_{y}}=\frac{450}{18.4 \times 10^{-6}}=24.46 \mathrm{MPa}$,
$\left|\sigma_{\max }\left(M_{z}\right)\right|=\frac{M_{z}}{W_{z}}=\frac{819}{143 \times 10^{-6}}=5.73 \mathrm{MPa}$.

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Fig. 2
Therefore,
$\sigma_{A}=-\sigma\left(N_{x}\right)-\sigma_{\max }\left(M_{y}\right)-\sigma_{\max }\left(M_{z}\right)=$
$=-4.27-24.46-5.73=-34.46 \mathrm{MPa}$,
$\sigma_{B}=-\sigma\left(N_{x}\right)-\sigma_{\max }\left(M_{y}\right)+\sigma_{\max }\left(M_{z}\right)=-4.27-24.46+5.73=-23.00 \mathrm{MPa}$,
$\sigma_{C}=-\sigma\left(N_{x}\right)+\sigma_{\max }\left(M_{y}\right)+\sigma_{\max }\left(M_{z}\right)=-4.27+24.46+5.73=25.92 \mathrm{MPa}$,
$\sigma_{D}=-\sigma\left(N_{x}\right)+\sigma_{\max }\left(M_{y}\right)-\sigma_{\max }\left(M_{z}\right)=-4.27+24.46-5.73=14.46 \mathrm{MPa}$.
3) Design the graph of stress distribution in $A B C D$ section in spatial cross section viewing with non-disturbed cross section for future geometrical analysis (see. Fig. 3).
4) Check the column strength:
(a) in tension, using the condition of strength
$\sigma_{\max _{t}} \leq[\sigma]_{t} \rightarrow \sigma_{\max _{t}}=\sigma_{C}=25.92 \mathrm{MPa}>20 \mathrm{MPa}$.
Conclusion: the column in non-strong in tension.
(b) in compression, using the condition of strength
$\left|\sigma_{\max _{c}}\right| \leq[\sigma]_{c} \rightarrow\left|\sigma_{\max _{c}}\right|=\left|\sigma_{A}\right|=34.46 \mathrm{MPa}<80 \mathrm{MPa}$.
Conclusion: the column is strong in compression.
General conclusion: the column is non-strong.

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Рис. 3
5) Determine the neutral axis position analytically. For this purpose, equate to zero the stress in an arbitrary point $K$ of $A B C D$ section. Let us select this point in the first quarter of the section (see Fig. 4).

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Fig. 4
$\sigma_{K}=-\frac{N_{x}}{A}+\frac{M_{y}}{I_{y}} z-\frac{M_{z}}{I_{z}} y=0$
(this is equation of the plane in $(\sigma, z, y)$ system of coordinates).
It may be rewritten as
$A z+B y+C=0$, where

$$
\begin{aligned}
& A=\frac{M_{y}}{I_{y}}=\frac{450}{82.6 \times 10^{-8}}=5.45 \times 10^{8} \\
& B=-\frac{M_{z}}{I_{z}}=-\frac{819}{1290 \times 10^{-8}}=-0.63 \times 10^{8} \\
& C=-\frac{N_{x}}{A}=-\frac{10 \times 10^{3}}{23.4 \times 10^{-4}}=-0.43 \times 10^{7}
\end{aligned}
$$

Find the segments which the neutral axis cuts on the coordinate axes:
(a) in $z=0$
$y^{*}=a_{y}=-\frac{C}{B}=-\frac{-0.43 \times 10^{7}}{-0.63 \times 10^{8}}=-0.683 \times 10^{-1} \mathrm{~m}=$

$$
=-68 \times 10^{-1} \mathrm{~m}=-68 \mathrm{~mm}
$$

(b) in $y=0$
$z^{*}=a_{z}=-\frac{C}{A}=-\frac{-0.43 \times 10^{7}}{5.45 \times 10^{8}}=+0.079 \times 10^{-1} \mathrm{~m}=+7.9 \times 10^{-3} \mathrm{~m}=+7.9 \mathrm{~mm}$.
These segments are shown in scaled cross section sketch on Fig. 4. Finally, the neutral axis is drawn through the segments tips.
6) Check the solution accuracy, correlating the neutral axis positions on Figs. 3 and 4.

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